

# Manufacturing Demonstration of a Large-scale, Multi-material Passenger Vehicle Sub-system

Project ID: mat243

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## OVERARCHING PROJECT GOAL

Implement a systems approach to redesign and manufacture a high-volume 2019 mid-size Honda SUV's glider system to achieve cost-effective and sustainable light-weighting through component consolidation, state-of-the-art optimization tools, multi-material joining methods, industry-standard manufacturing processes and recycling technologies while meeting or exceeding baseline performance.

### TIMELINE

- Start: October 1<sup>st</sup>, 2021
- End: December 31<sup>st</sup>, 2024
- Level of Completion: 15 %

### BUDGET

- Total Project Funds: \$ 11.5 Million
- \$ 5.75 Million – DOE
  - \$ 5.75 Million – Cost share

## BARRIERS IN ACHIEVING PROJECT GOAL

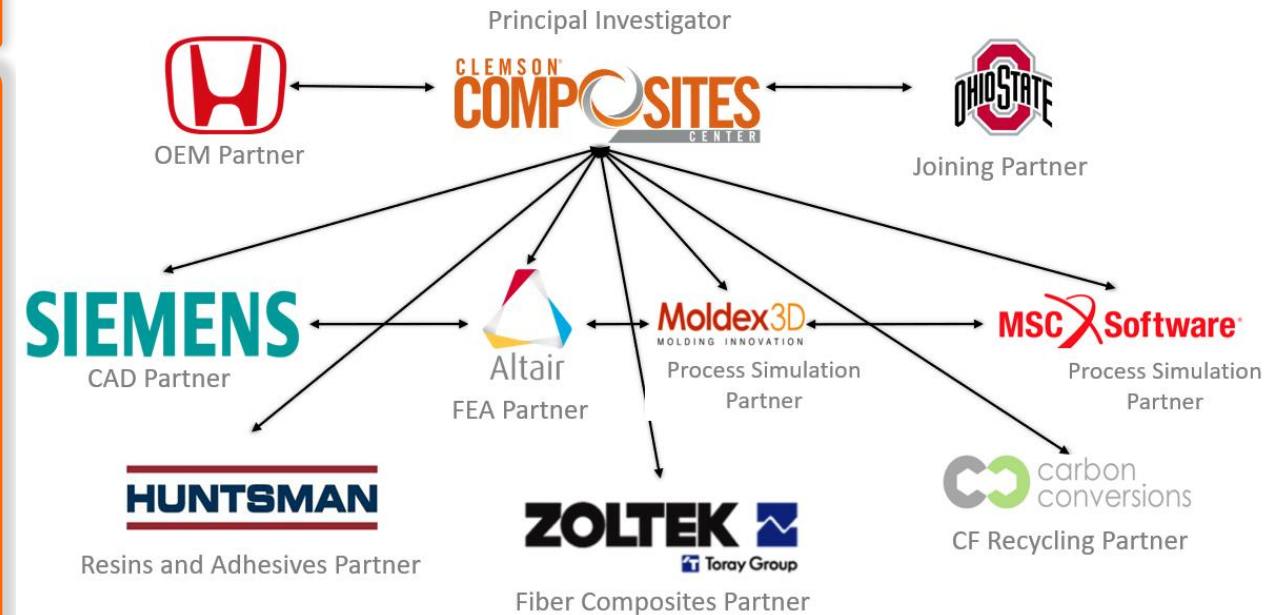
#### Cost-performance tradeoff

- Identifying the trade-off between composites and traditional metals for cost-effective redesign
- Using composites while enabling fast manufacturing cycle times and preserving OEM's existing joining & assembly infrastructure
- Achieving cost and light-weighting targets while meeting or exceeding baseline performance targets
- Developing and integrating cost, sustainability, and performance predictive tools to identify suitable material systems and designs.

#### Material – Process – Property Integration

- Testing & characterization for material aging data cards development
- Development of process data cards through manufacturing process experimentation – simulation loop.
- Development and deployment of coupled ply-forming and resin infiltration model for Wet Compression Molding (WCM)
- Developing compatibility between transition joints and WCM

## PROJECT PARTNERS





# Project Relevance

1. **Achieve a 160 lbs. (73 kg) weight reduction**
  - Aligns with U.S DRIVE Roadmap to enable weight reduction
2. **Zero compromise on performance targets**
  - No compromise in crash performance, NVH, durability, strength and stiffness
  - No compromise in the fit & packaging of other sub-systems
3. **Cost increment limited to \$5 per pound (.453 kg)**
  - Allowable increase of \$800 per glider system
4. **Scalability**
  - Annual production of 200,000 vehicles
  - Preservation/simplification of the OEM's factory assembly process
5. **Recyclability**
  - Utilize recycled carbon fiber to enable sustainable light-weighting



## **BUDGET PERIOD 1 (1st October 2021 to 31st December 2022)**

- ✓ Design Requirements completed (Q4 2021)
- ✓ Conceptual Designs completed (Q1 2022)
- ❖ Metal-CF Transition Joint Machine Adaptations completed (Q4 2022)
- ❖ Metal-CF transition joint modeling completed (Q4 2022)
- ❖ Down-selected Conceptual Designs meeting performance criteria (Q4 2022)

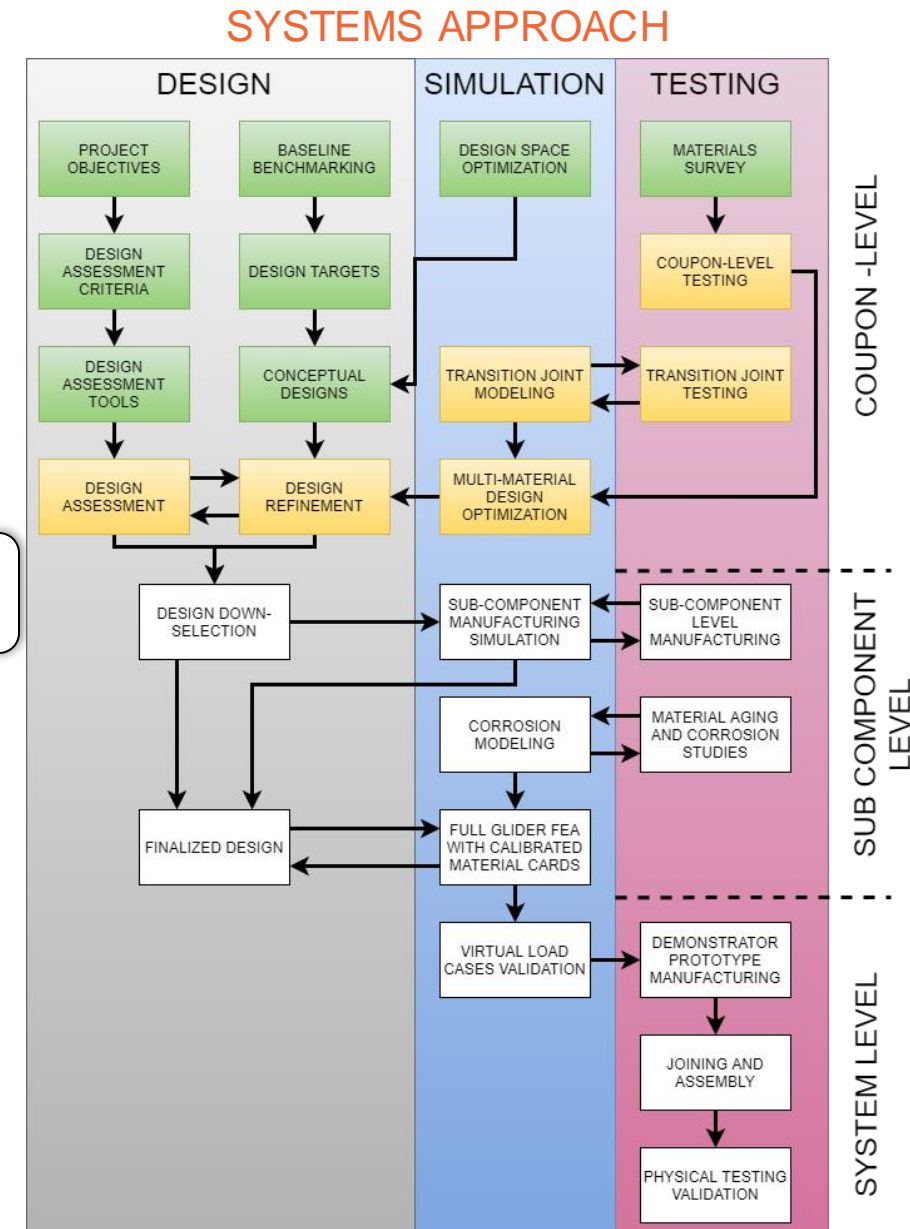
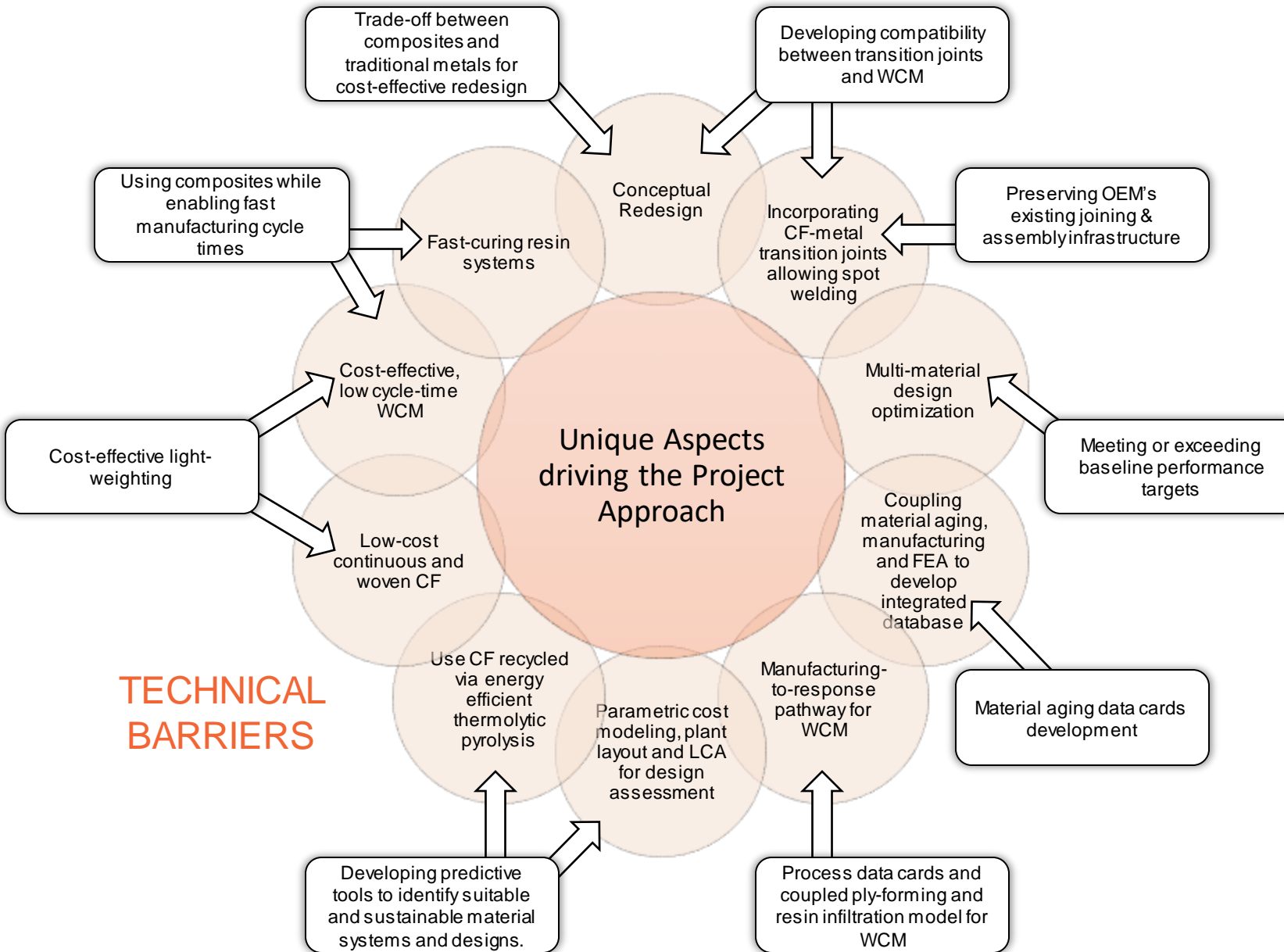
## **BUDGET PERIOD 2 (1st January 2023 to 31st December 2023)**

- ☐ Metal-CF transition joints verified to meet performance criteria (Q3 2023)
- ☐ Corrosion and life-time prediction models validated (Q4 2023)
- ☐ Performance prediction validated (Q4 2023)
- ☐ Mock-up sub-system verified to meet fit and integration criteria (Q3 2023)

## **BUDGET PERIOD 3 (1st January 2024 to 31st December 2024)**

- ☐ Factory layout completed (Q4 2024)
- ☐ Glider recycling strategy complete (Q4 2024)
- ☐ Glider assembly complete (Q2 2024)
- ☐ Glider performance tests complete (Q3 2024)

# Project Approach



# Progress: Concept Development

## Materials

- Baseline Steel, Aluminum
- FRP thermosets (CF, GF, Basalt, Kevlar)
- FRP thermoplastics (CF, GF, Basalt, Kevlar)

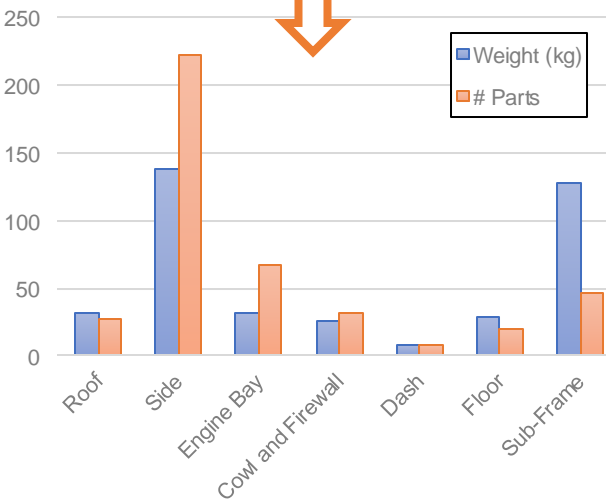
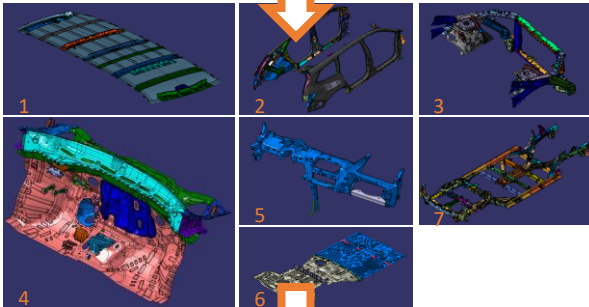
## Manufacturing Processes

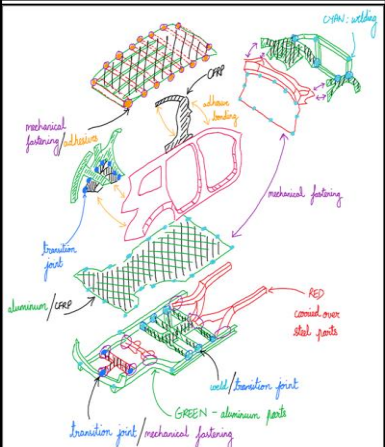
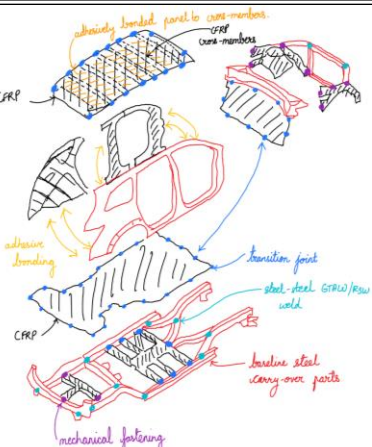
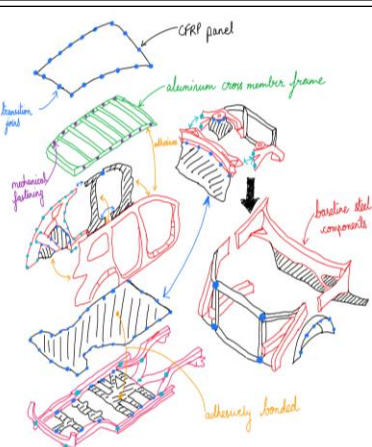
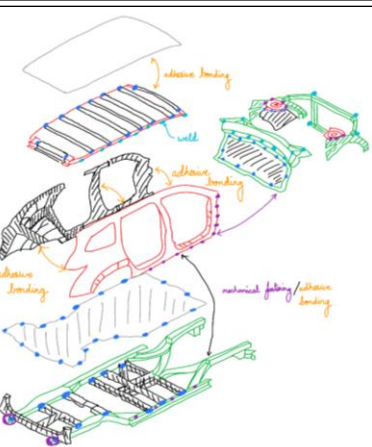
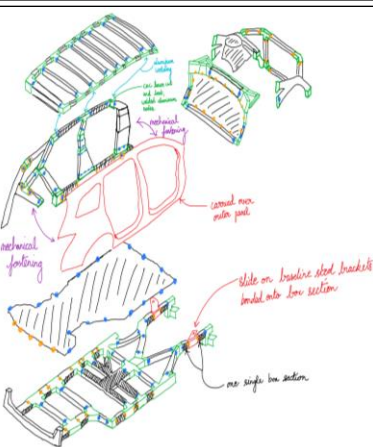
- Standard steel/aluminum sections
- HPRTM, WCM, SMC
- Thermoplastic Forming
- Injection Molding



## Joining Processes

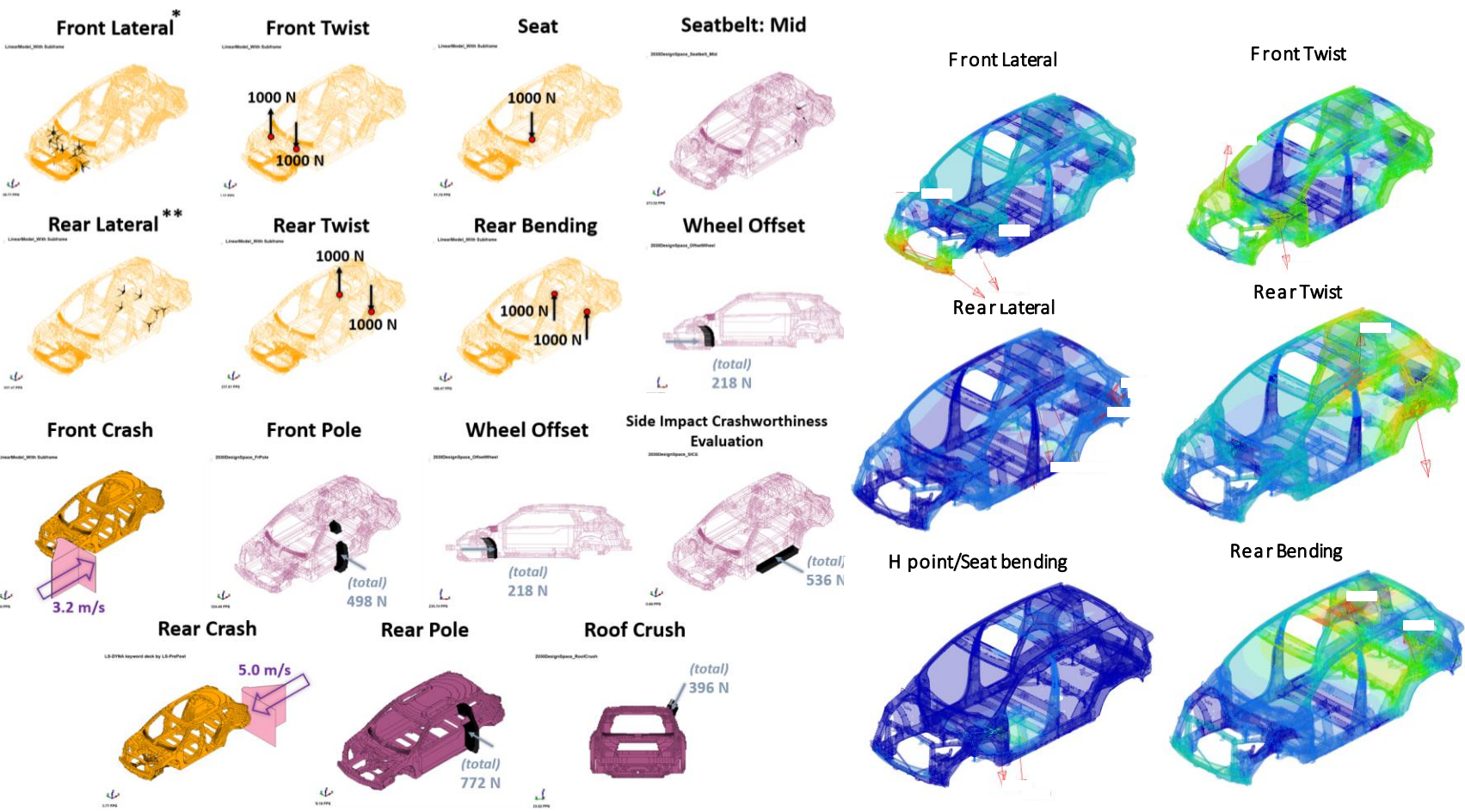
- Mechanical Fastening
- Welding – RSW, MIG
- Adhesives – Adhesive bonding
- Transition joints



Concept 1	Concept 2	Concept 3	Concept 4	Concept 5
				
<ul style="list-style-type: none"> <li>• Carry-over steel parts</li> <li>• major replacement with aluminum</li> <li>• selective parts consolidation with composites</li> </ul>	<ul style="list-style-type: none"> <li>• Carry-over steel parts</li> <li>• major parts consolidation with composites</li> </ul>	<ul style="list-style-type: none"> <li>• CFRP-steel construction with transition joints</li> <li>• CFRP-aluminum-steel transition joints for roof</li> <li>• Subframe outer members and mid-rails retained</li> <li>• Cross members and floor consolidated into a composite design</li> </ul>	<ul style="list-style-type: none"> <li>• Welded Aluminum sections as frames for composite panels</li> <li>• use of basalt fibers/GFRP for panels</li> <li>• Side panel and stiffener sub-assembly consolidated into single composite design</li> </ul>	<ul style="list-style-type: none"> <li>• Spaceframe with metal nodes</li> <li>• Adhesive bonding</li> <li>• RSW + transition joints leveraged</li> <li>• Favorable for prototype technology demonstrator glider</li> </ul>
<ul style="list-style-type: none"> <li>• Weight Reduction Potential: 17.4%</li> </ul>	<ul style="list-style-type: none"> <li>• Weight Reduction Potential: 27 %</li> </ul>	<ul style="list-style-type: none"> <li>• Weight Reduction Potential: 30%</li> </ul>	<ul style="list-style-type: none"> <li>• Weight Reduction Potential : 43%</li> </ul>	<ul style="list-style-type: none"> <li>• Weight Reduction Potential: 39%</li> </ul>
<ul style="list-style-type: none"> <li>• Parts Consolidation Potential: 21.75%</li> </ul>	<ul style="list-style-type: none"> <li>• Parts Consolidation Potential: 32.8%</li> </ul>	<ul style="list-style-type: none"> <li>• Parts Consolidation Potential: 30.5%</li> </ul>	<ul style="list-style-type: none"> <li>• Parts Consolidation Potential: 39.5%</li> </ul>	<ul style="list-style-type: none"> <li>• Parts Consolidation Potential: 41%</li> </ul>
<ul style="list-style-type: none"> <li>• OEM Assembly line compatibility: <b>LOW</b></li> </ul>	<ul style="list-style-type: none"> <li>• OEM Assembly line compatibility: <b>HIGH</b></li> </ul>	<ul style="list-style-type: none"> <li>• OEM Assembly line compatibility: <b>MODERATE</b></li> </ul>	<ul style="list-style-type: none"> <li>• OEM Assembly line compatibility: <b>MODERATE to LOW</b></li> </ul>	<ul style="list-style-type: none"> <li>• OEM Assembly line compatibility: <b>LOW</b></li> </ul>



# Progress: Structural Performance



A summary of load cases and displacement contour plots for the considered baseline model.

Description of Load Cases considered

	Load Case	Description
1	Front Lateral	Lateral loading at front suspension points
2	Rear Lateral	Lateral loading at rear suspension points
3	Front Twist	Torsional loading at front suspension points
4	Rear Twist	Torsional loading at rear suspension points
5	Rear Bend	Bending load at rear suspension points
6	H point loading	Loading at seat mounting points
7	Front Crash	Small overlap front impact
8	Rear Crash	Small overlap rear impact
9	Side Pole Impact	Side pole loading near front/rear of vehicle
10	Roof Crush	Loading on the roof
11	Wheel Offset	Loading on front left wheel well
12	SICE	Side Impact Crashworthiness Evaluation (IIHS)

**\*Static Load Cases. (solved)**

**\*Dynamic Load Cases. (ongoing)**

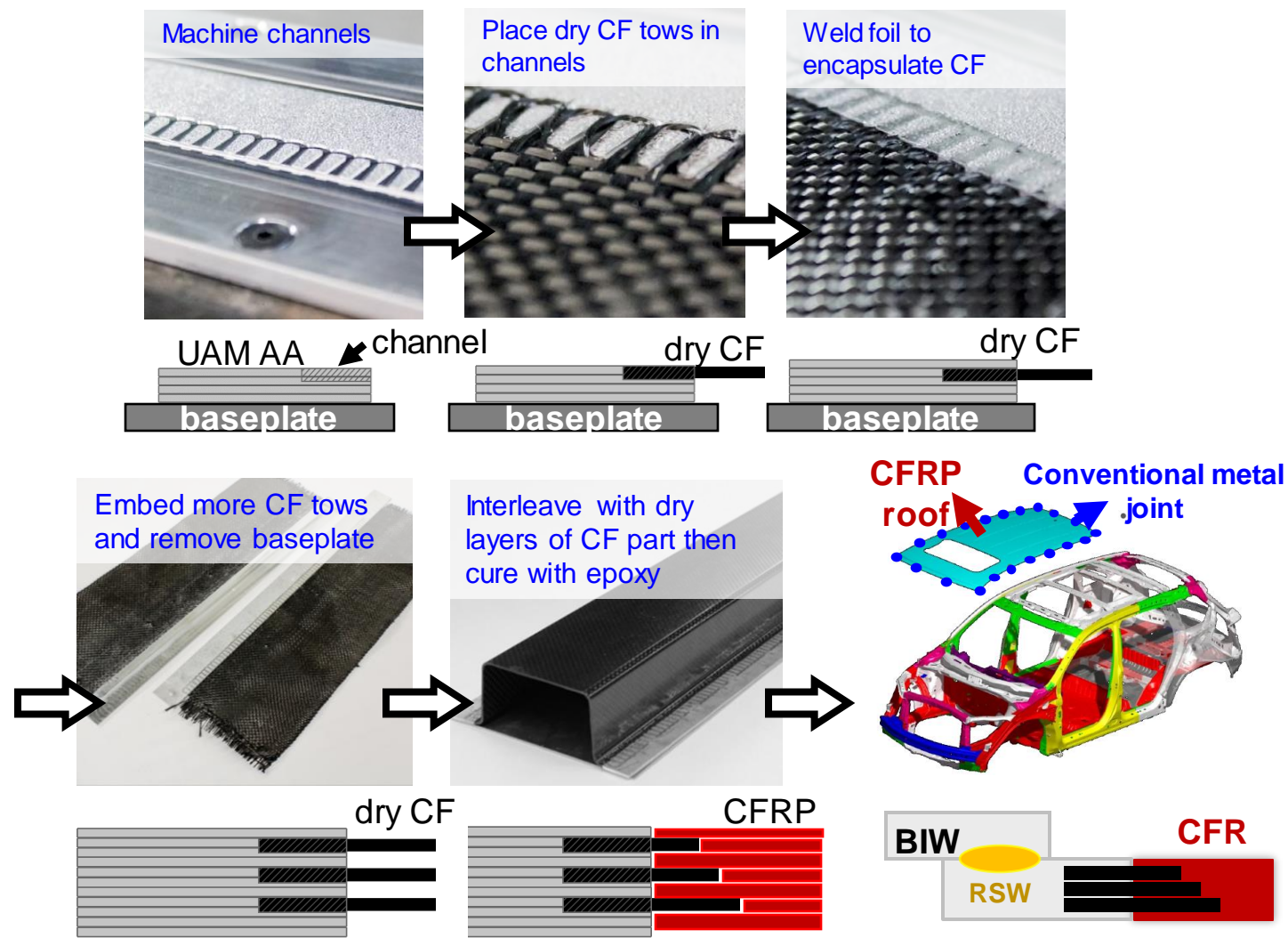
# Progress: UAM Process



OSU's UAM system with automated foil feeder and CNC

<b>Tensile strength</b>	CF: 102.3 MPa Al: 129.5 MPa
<b>Cross-tensile strength</b>	2318 N
<b>Corrosion cycling</b>	No degradation after 120 cycles
<b>Axial crush</b>	+66% EA* +97% efficiency
<b>Four-point bending</b>	+86% EA +83% efficiency
<b>Torsion</b>	+48% EA +13% peak load

\* EA: energy absorption



Research on material combinations, design, and manufacturing integration to achieve light-weighting and cost targets while meeting strength, corrosion, and safety requirements



# Collaborations

## ACADEMIA PARTNERS



## OEM PARTNER



## GLOBAL N-TIER PARTNERS



## MATERIALS & RECYCLING PARTNERS



- Task 1.3: Cost, cycle times, plant layout and LCA
  - Assess, refine and down-select conceptual designs
  - Identifying process flows for LCA
- Task 1.4: End of life recycling
  - Identify opportunities to use CCI's recycled CF preforms in proposed conceptual designs
  - Incorporate CCI's process data into LCA
- Task 1.5: Transition Joint Development
  - Develop model for transition joints and incorporate into topology optimization
  - Develop material card for use in topology optimization using results from coupon-level transition joint model.
  - Complete UAM joint fabrication machine adaptations to meet cycle time requirements



❑ Milestone: Transition joint modeling complete (Q4 2022)



❑ Milestone: UAM machine adaptations complete (Q4 2022)

- Task 1.6: Multi-material Glider Optimization
  - ✓ Baseline FEA completed and results evaluated to establish complete set of design requirements/targets
  - Single material topology optimization
  - Multi-material topology optimization



❑ Milestone: Conceptual design meet performance criteria (Q4 2022)

# Summary



## Task 1.1 Glider design requirements - Complete

- ✓ Milestone: Design Requirements complete

## Task 1.2 Systems design

- ✓ Milestone: Conceptual Designs complete
- Design refinement and down-selection in-progress

## Task 1.3 Cycle time assessment, cost modeling and LCA

- Methodology for Cost Model was developed
- Design concepts assessment in terms of manufacturability and cost in-progress.

## Task 1.4 End of life recycling

- Initial data from carbon conversions being assessed

## Task 1.5 Metal-CF transition structure

- Coupon-level modeling using existing materials in progress
- UAM metal-CFRP transition joint fabrication process and system being developed to speed up fabrication. Beginning research on transition materials and design for specific joining applications.

## Task 1.6 Multi material design optimization

- Baseline FEA completed and results evaluated to establish complete set of design requirements/targets
- Single material topology optimization in progress





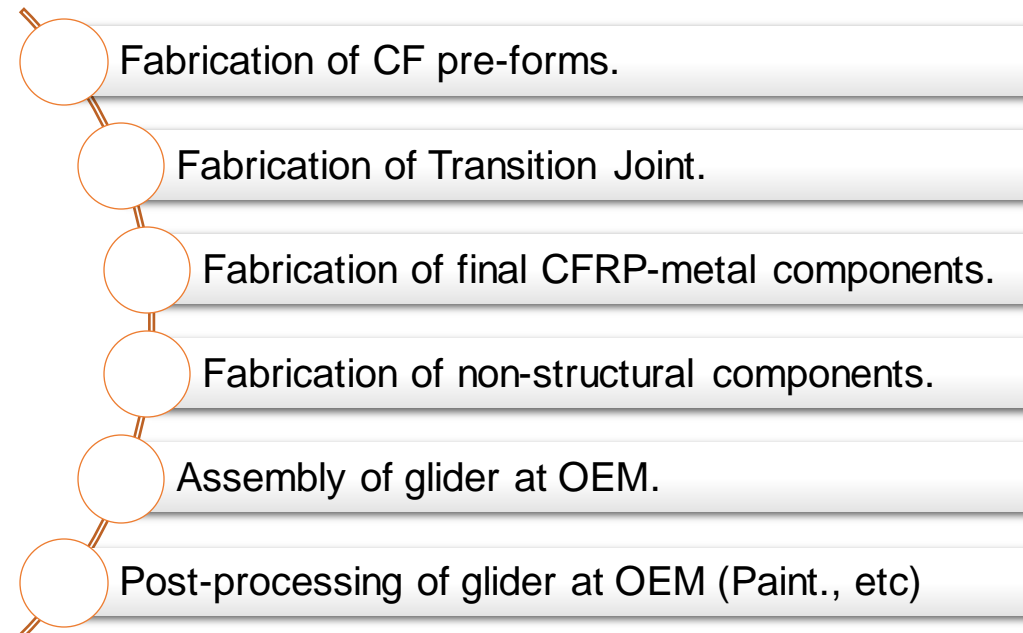
# Technical Back Up Slides

## › **Objective:**

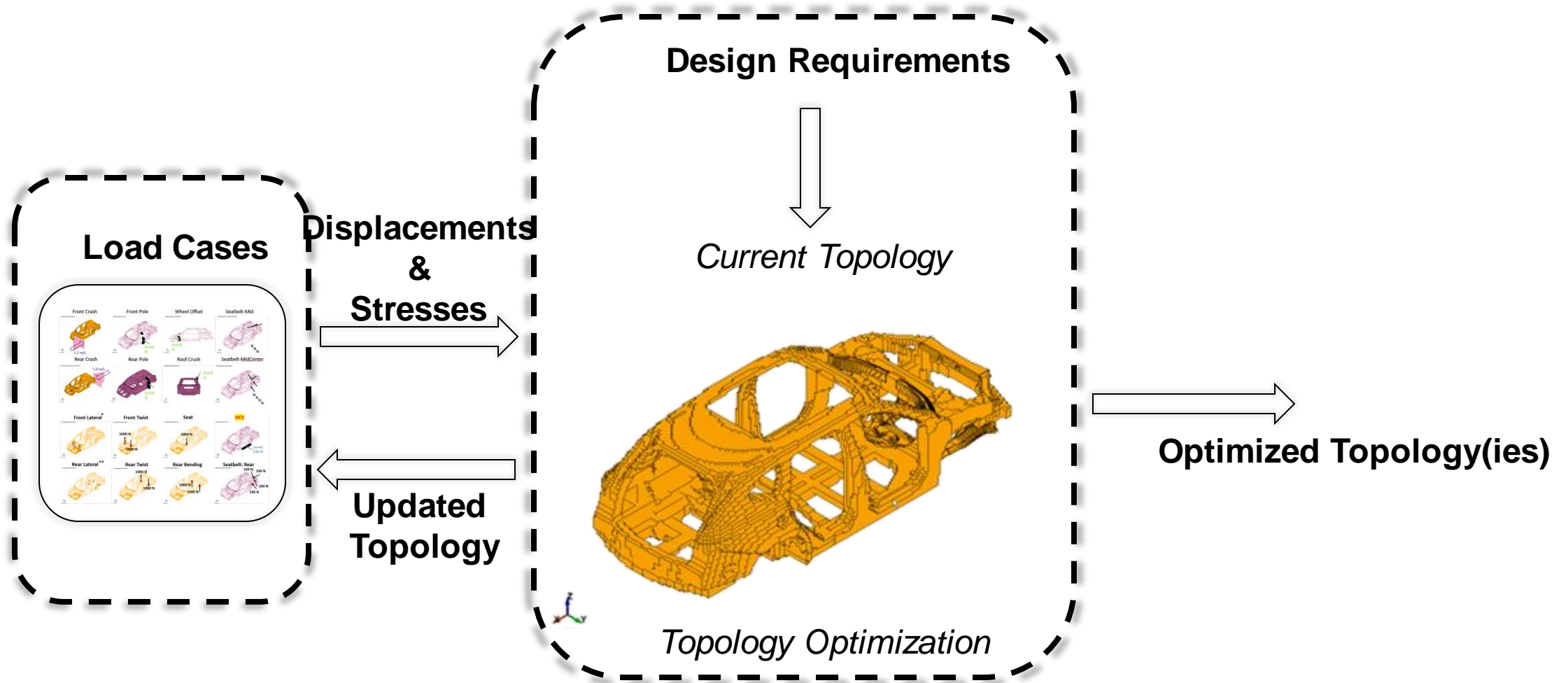
- Develop a cost model that includes the cost of material, equipment, labor, energy consumed for the fabrication of all structural and non-structural parts of the glider.

## › **Scope:**

- Includes all processes, both at OEM and offline.



# Topology Optimization Workflow

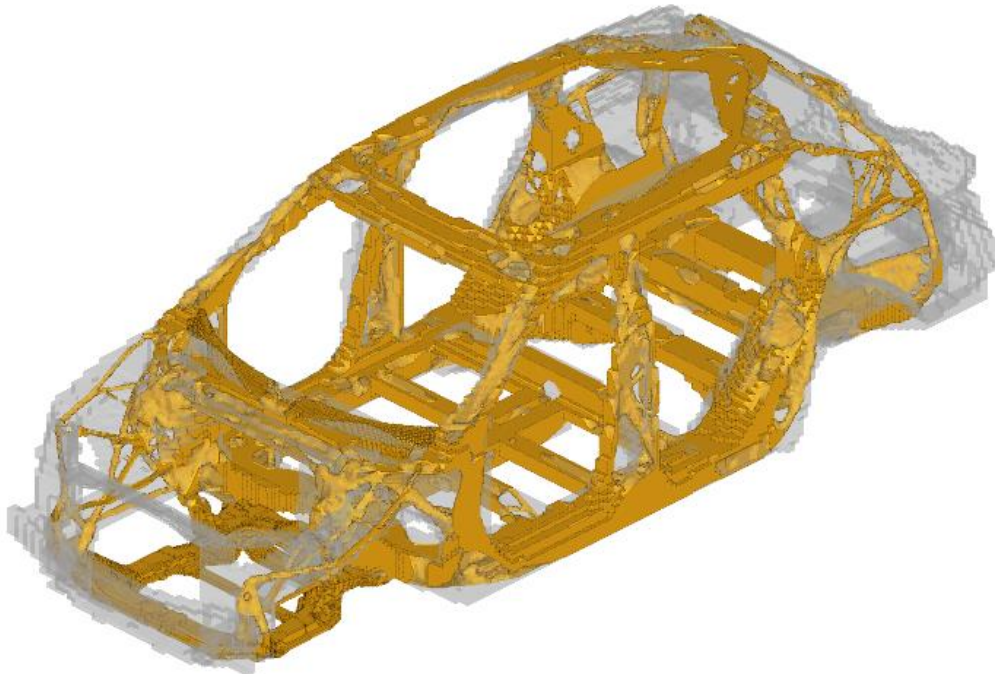
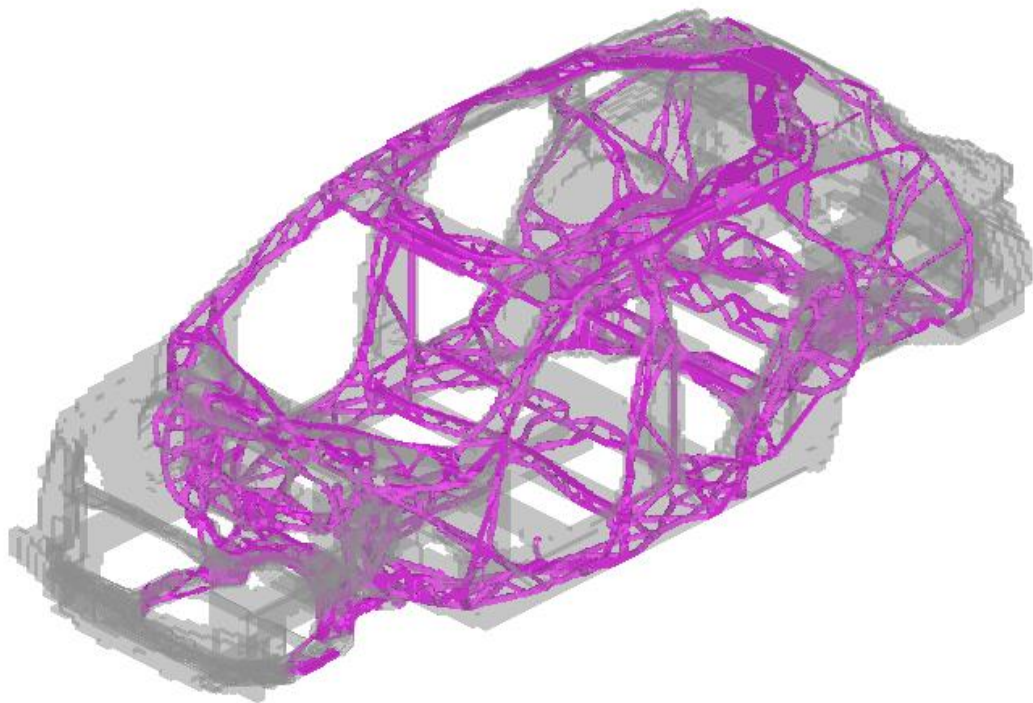


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# Single Material Topology Optimization Results

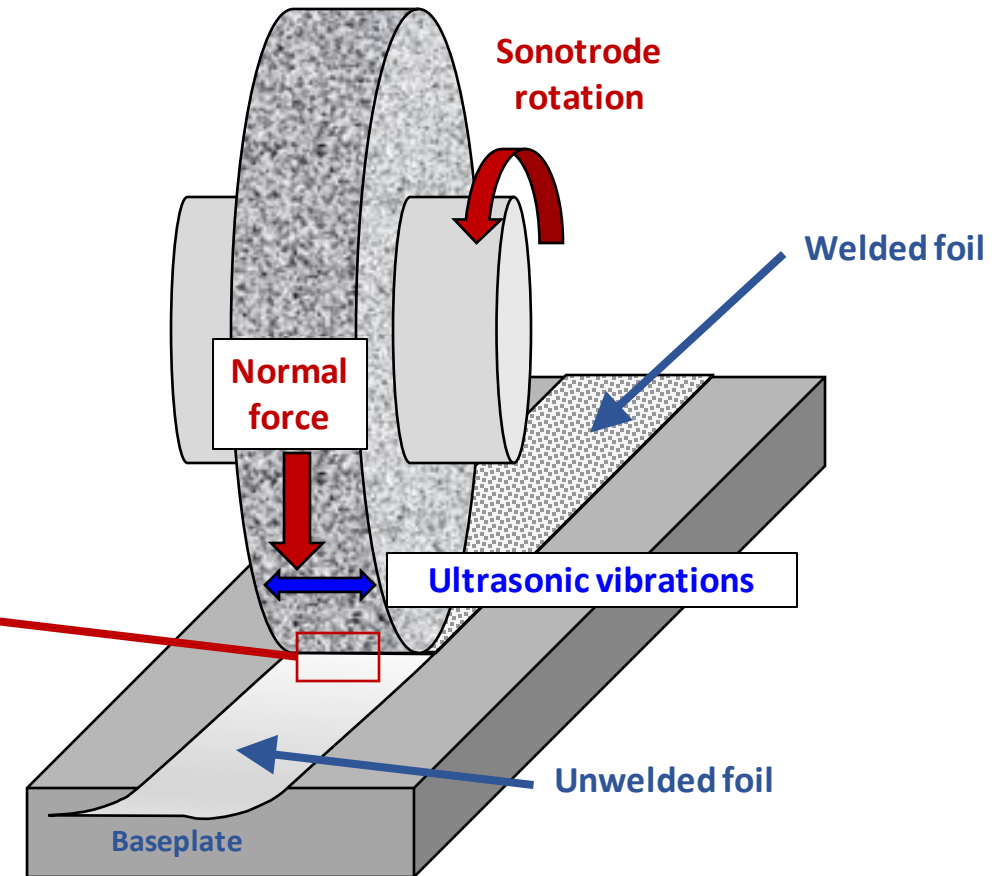
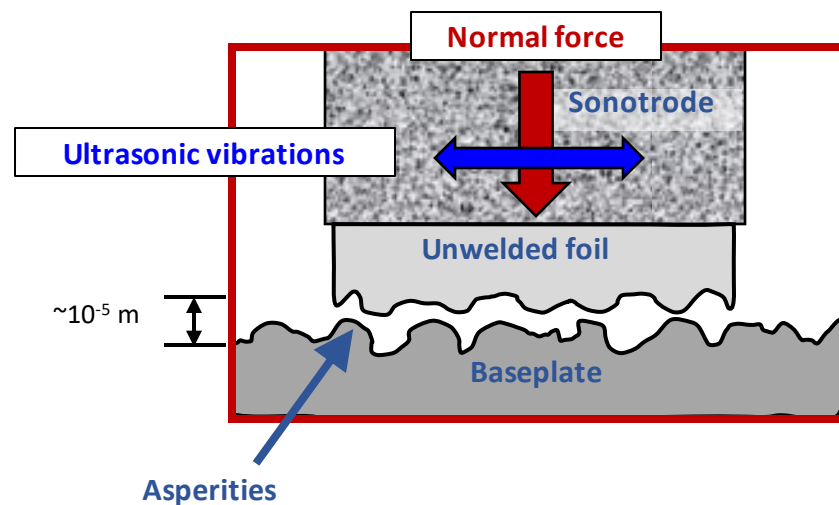
\*Static cases only



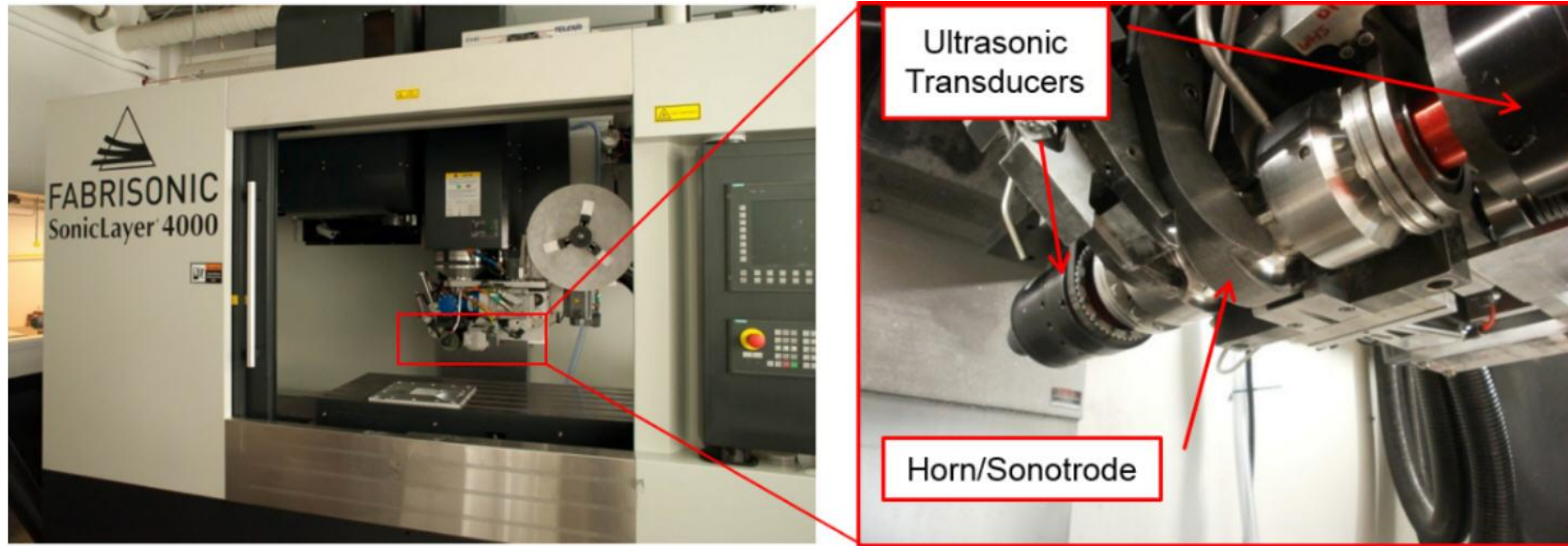
Objective	Min V	Min $\Sigma$ compliance
Subject To	<ul style="list-style-type: none"><li>• Max stress constraint</li><li>• Baseline compliance constraint</li></ul>	<ul style="list-style-type: none"><li>• Max stress constraint</li><li>• 25% (of design space) volume constraint</li></ul>
Material	Steel	Steel

# Ultrasonic Additive Manufacturing (UAM) - Process

- Ultrasonic additive manufacturing uses solid-state ultrasonic metal welding to create metal parts
- Dispersal of oxide layer and collapsing of asperities lead to metallic bonding
- Low-temperature process
- Unprecedented opportunity to embed temperature sensitive materials in metals
- Combined with CNC and laser can create near net shape parts



# Ultrasonic Additive Manufacturing (UAM) - System



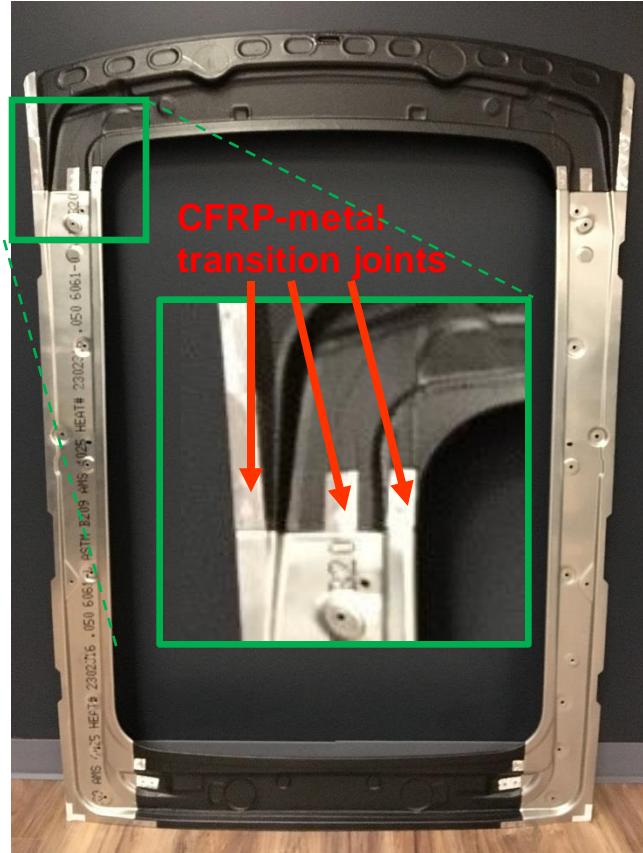
## 9 kW SonicLayer 4000 UAM system

- 3-axis CNC mill with 25 HP, 8000 RPM spindle
- 1 m × 1 m × 0.6 m work volume
- 9 kW piezoelectric dual weld head and sonotrode for welding a variety of alloys including Al, Ti, and stainless steel
- 40 W integrated laser for micromachining, surface modification, and augmented embedding
- Automated feeder for metal foil
- Instrumented to record sensor data and input power during welding

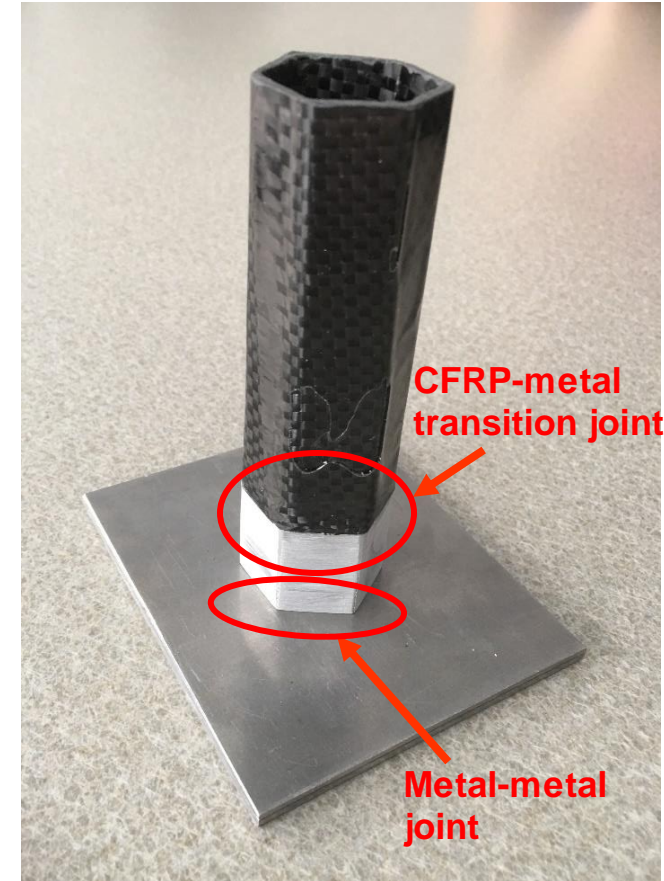


# Previous Results and Feasibility – Demonstration Parts

Since **dry fibers** are embedded in a **choice of metals**, virtually any CFRP part can be laid up with **integrated metal tabs, ribs, or attachment points** that can be joined to other metal structures using **traditional metal-to-metal joining methods**

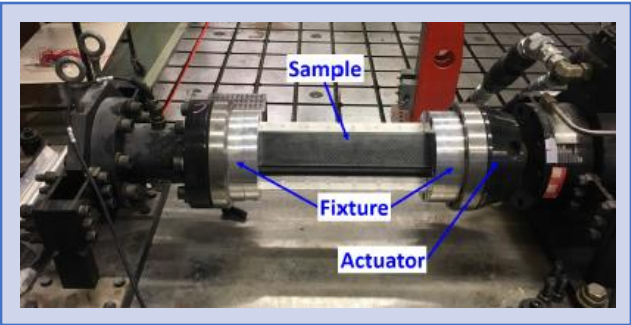
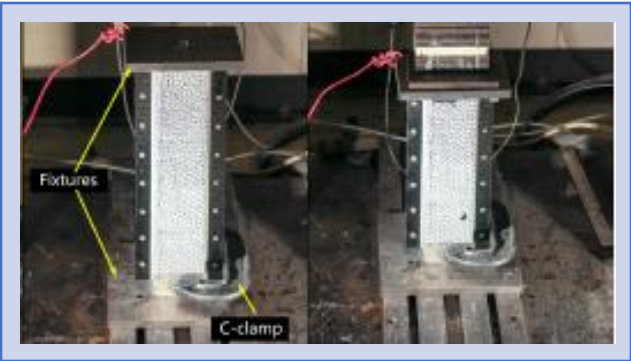
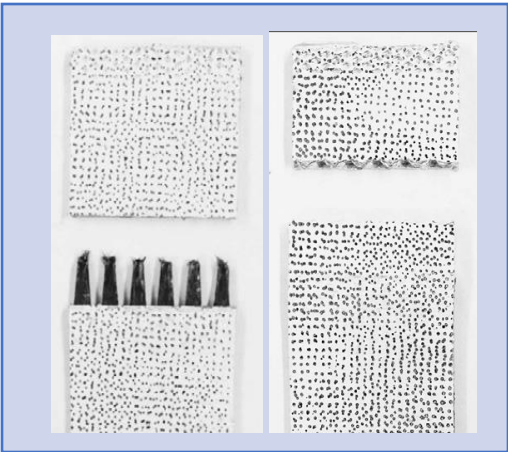


**Aluminum and CFRP hybrid roof stiffener with UAM CFRP-metal transitions**

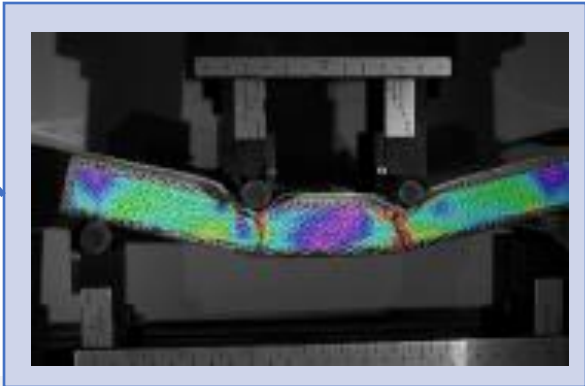
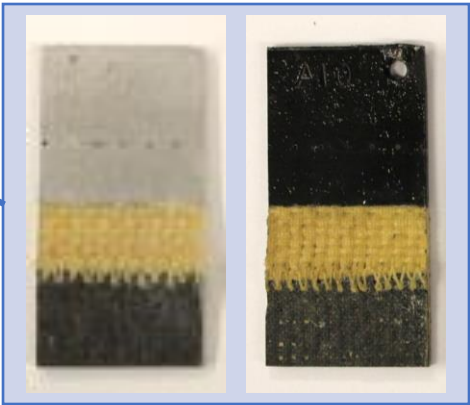
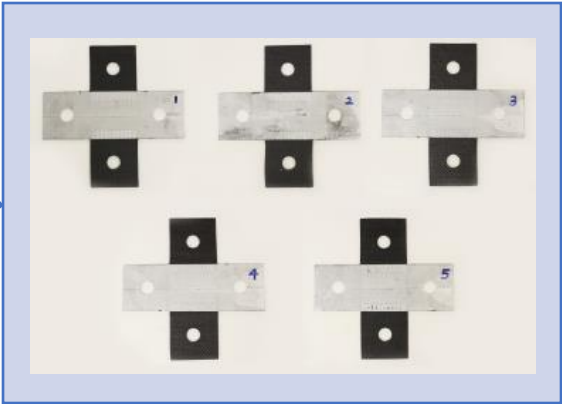


**CFRP hollow rod with integral aluminum end connector, joined via UAM CFRP-metal transitions. End connector is attached to a metal base with a conventional metal-metal joint.**

# Previous Results and Feasibility – Key Test Results



<b>Tensile strength</b>	CF: 102.3 MPa Al: 129.5 MPa
<b>Cross-tensile strength</b>	2318 N
<b>Corrosion cycling</b>	No degradation after 120 cycles
<b>Axial crush</b>	+66% EA* +97% efficiency
<b>Four-point bending</b>	+86% EA +83% efficiency
<b>Torsion</b>	+48% EA +13% peak load



\* EA: energy absorption

## **Task 5. Metal-CF transition structure development (M4-M24)**

Perform detailed [metal–CF transition joint development](#) for [fiber/fabric form choices](#), [performance](#) (tensile shear and cross-tension strengths), [manufacturing approaches](#), [joining and assembly options](#) (spot welding or adhesives) and [glider assembly](#) to meet performance weight, cycle time, and cost goals

### ***Task 5.1: UAM process adaptation (M4-M6)***

- Adapt the UAM fabrication process to [accommodate making steel-buffer-CF and aluminum-buffer-CF transition joints for glider sub-system structures](#) over the course of this research program, including:
  - Designing and building a [larger base fixture](#)
  - Installing a [high-speed spindle](#)
  - Fabricating a [tool to facilitate placement of loops in the machined channels](#).
- [Evaluate cost-effective fabrics](#) such as glass fibers (E, S) and basalt fibers as buffer materials, compared with Kevlar

### ***Task 5.2: UAM process conditions (M7-M9)***

- Conduct welding trials and shear strength testing to [identify good welding parameters](#)
  - Utilize [automotive-grade aluminum and low carbon steel](#) that are compatible with automotive RSW processes
- [Build and test steel-buffer transitions](#) to assess weld quality and mechanical properties

### ***Task 5.3. Fabrication and characterization of metal-CF transitions (M10–M24)***

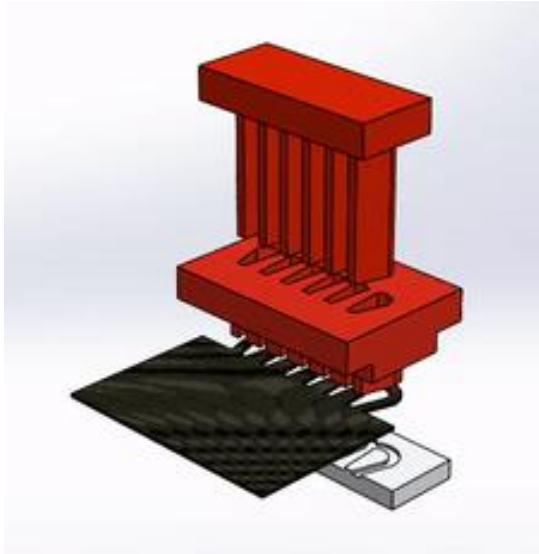
- [Fabricate samples for testing](#) according to the design developed in subtask 6.2
- [Test and characterize samples](#) according to standard automotive company tests used by the automotive industry
  - Tensile shear strength (TSS)
  - Cross-tension strength (CTS)
- [Representative structures or components](#) will be [fabricated and tested](#) based on the simulated loading conditions from glider sub-systems and assembly-level simulations. These validation tests will help ensure that the transitions will perform as intended for multi-material vehicle gliders.



# Progress: UAM Process

## Ultrasonic additive manufacturing (UAM) system modifications to speed up fabrication of metal-CFRP transitions

Henninger Type-832 speed increaser selected (36,000 RPM continuous operation)



Fiber alignment device prototyped for faster placement of tows

